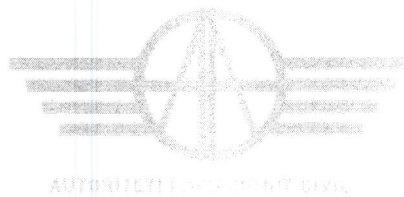




REPUBLIC OF ALBANIA



ALBANIAN CIVIL AVIATION AUTHORITY

AIRWORTHINESS DIRECTIVE

ACAA-DFS-AD-No.038

Issue: 01, Revision 00

Date: 28.04.2025

Approved by:

Maksim Et'hemaj

Executive Director of Albanian Civil Aviation Authority



0.1 Record of Amendments

The table below describes the dates and reason for the different amendments of the current procedure. A vertical black line on the left-hand side of the page identify the changes with the previous version.

Issue No.	Revision No.	Date	Amended by	Reason
01	00	28.04.2025	SAW	Initial Issue

0.2 Revision table

Page #.	Issue No.	Revision No.	Date	Edited by

1. Name of the AD:

ATA 25 – Equipment / Furnishings – Galleys – Inspection

2. Issued and Effective Dates:

Issued: 28 March 2025

Effective Date: 11 April 2025

Revision:

Supersedure:

ACAA-DFS-AD-No.005

3. Full List of Aircraft Affected:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252N, A321-252NX, A321-253N, A321-253NX, A321-271N, A321-271NX, A321-272N and A321-272NX aeroplanes, all manufacturer serial numbers.

4. Definitions:

For the purpose of this AD, the following definitions apply:

Affected part: Forward-facing galleys, having a Part Number (P/N) as listed in Appendix 1 of this AD.

The inspection SB: Airbus Service Bulletin (SB) A320-25-1BVS or Airbus SB A320-25-1BVT, as applicable.

The modification SB: Airbus SB A320-25-1CBN or A320-25-1CBP, as applicable.

Aeroplane date of manufacture: The date of transfer of title (ownership) of the aeroplane at the time of first delivery to an operator, which is referenced in Airbus documentation.

Groups:

Group 1 aeroplanes are those that have an affected part installed.

Group 2 aeroplanes are those equipped with a galley which has been initially installed as an affected part on an aeroplane in production or in service in accordance with Airbus service instructions, and which following that installation, has been re-identified with a non-affected P/N without having the modification SB embodied. The identification of the galleys installed during production can be found in the Aircraft Inspection Report (AIR).

Group 3 aeroplanes are those which are not Group 2 aeroplanes and do not have an affected part installed.

5. Reason:

Occurrences have been reported of finding damage during inspection of certain galleys, including delamination of work deck as well as corroded and cracked retainer blocks. This condition, if not detected and corrected, could affect the galley's capability to hold the trolley under emergency landing loads, which could lead to trolley detachment, possibly resulting in blocking of an escape path during an emergency exit.

To address this potential unsafe condition, Airbus issued the inspection SB, providing instructions to inspect the affected parts, and EASA issued AD 2021-0183 to require repetitive general visual inspections (GVI) of each affected part and, depending on findings, accomplishment of applicable corrective action(s).

After AD 2021-0183 was issued, it was determined that, following accomplishment of a certain repair, subsequent inspection of the repaired area could be deferred beyond the established inspection interval. Additionally, typographical errors were identified in Appendix 1 of that AD. Prompted by this development EASA issued EASA AD 2021-0183R1, accordingly.

After EASA AD 2021-0183R1 was issued, Airbus and the galley manufacturer developed a modification to restore the design integrity of the affected galleys, and Airbus issued the modification SB, as defined in this AD, providing instructions for in-service galley modification. It was also identified that some affected galley P/N were missing in the Appendix 1 of EASA AD 2021-0183R1, while others could be removed from that list. Consequently, EASA issued accordingly AD 2024-0038.

Since EASA AD 2024-0038 was issued, it was identified that galleys having P/N 601891-006801, P/N 601891-003701 and P/N 601891-010001 were missing in the Appendix 1 of that AD, and it was identified that galleys having P/N 6019A3-000101 are not affected by the unsafe condition addressed by that AD.

For the reasons described above, this AD retains the requirements of EASA AD 2024-0038, which is superseded and includes an updated list of affected parts.

The scope of this AD is limited to affected parts under the responsibility of Airbus as Design Organisation Approval holder. EASA considers separate mandatory actions for galleys modified or installed under the responsibility of design approval holders other than Airbus (see also Remark 4 of this AD for reporting).

6. Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Inspection(s):

- (1) For Group 1 aeroplanes: Within the compliance time as defined in Table 1 of this AD, as applicable, and, thereafter, at intervals not exceeding 6 months, accomplish a GVI of each affected part in accordance with the instructions of the inspection SB (see Note 1 of this AD).

Table 1 – Initial Inspection Compliance Time for Group 1 Aeroplanes

Group 1 Aeroplanes Configuration	Compliance Time
Group 1 aeroplanes, except the ones below	36 months since first installation of an affected part on an aeroplane, or within 12 months after 18 August 2021 [the effective date of the EASA AD 2021-0183 at original issue], whichever occurs later

Equipped with an affected part having P/N 6018A7-000101 or P/N 6018C1-000101	Within 12 months after 19 February 2024 [the effective date of the EASA AD 2024-0038]
Equipped with an affected part having P/N 601891-006801 or P/N 601891-003701 or P/N 601891-010001	Within 12 months after the effective date of this AD

Note 1: If the date of first installation of an affected part on an aeroplane is unknown, the aeroplane date of manufacture can be used instead.

- (2) For Group 2 aeroplanes: Within the compliance time as defined in Table 2 of this AD, as applicable, contact the galley manufacturer, Airbus, or the design approval holder responsible for the re-identification of the galley, as applicable, for further instructions and accomplish those instructions accordingly. The instructions must be approved by EASA or under the authority of an EASA Design Organisation Approval holder.

Table 2 – Compliance Time for Group 2 Aeroplanes

Group 2 Aeroplanes Configuration	Compliance Time
Group 2 aeroplanes, except the ones below	12 months after 19 February 2024 [the effective date of EASA AD 2024-0038]
Group 2 aeroplanes equipped with a galley which has been initially installed as P/N 601891-006801 or P/N 601891-003701 or P/N 601891-010001	Within 12 months after the effective date of this AD

Corrective Action(s):

- (3) If, during any GVI as required by paragraph (1) of this AD, discrepancies are detected on an affected part, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the inspection SB.

Part(s) Installation:

- (4) For Group 1, Group 2 and Group 3 aeroplanes: From the effective date of this AD, installation of an affected part is allowed, provided that, following installation, it is inspected and, depending on findings, corrected, as required by this AD.

GVI after Repair:

- (5) For any panel and/or retainer block of an affected part on which an intermediate repair has been accomplished in accordance with the instructions of the inspection SB, the next GVI as required by paragraph (1) of this AD can be deferred until 36 months after that repair. Thereafter, inspect that panel and/or retainer block at intervals not exceeding 6 months, as required by paragraph (1) of this AD.

Terminating Action:

- (6) Modification of an affected part on an aeroplane, accomplished in accordance with the instructions of the modification SB, constitutes terminating action for the initial and repetitive inspections as required by paragraphs (1) and (5) of this AD, as applicable, for that galley.

- (7) Modification of all affected parts on an aeroplane, accomplished in accordance with the instructions of the modification SB, constitutes terminating action for the initial and repetitive inspections as required by paragraphs (1) and (5) of this AD, as applicable, for that aeroplane, provided that no affected parts are re-installed on that aeroplane after that modification.

Acceptable Method:

- (8) For Group 1 and Group 2 aeroplanes: Replacement of a forward-facing galley of an aeroplane with a galley which has been modified in accordance with approved instructions, which include the accomplishment of the post-modification SB instructions, constitutes terminating action for the initial and repetitive inspections as required by paragraphs (1), (2) and (5) of this AD, as applicable, for that galley of that aeroplane.

7. Ref. Publications:

Airbus SB A320-25-1BVS original issue dated 18 December 2020, or Revision 1 dated 17 June 2022, or Revision 2 dated 10 August 2022.

Airbus SB A320-25-1BVT original issue dated 18 December 2020, or Revision 1 dated 17 June 2022.

Airbus SB A320-25-1CBN original issue dated 12 August 2022, or Revision 1 dated 20 March 2023, or Revision 02 dated 30 August 2024.

Airbus SB A320-25-1CBP original issue dated 12 August 2022, or Revision 1 dated 20 March 2023, or Revision 02 dated 30 August 2024.

The use of later approved revisions of the above-mentioned documents is acceptable for compliance with the requirements of this AD.

8. Remarks:

- (1) If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
- (2) This AD was posted on 03 February 2025 as PAD 25-025 for consultation until 03 March 2025. The Comment Response Document can be found in the EASA Safety Publications Tool, in the compressed (zipped) file attached to the record for this AD.
- (3) Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
- (4) Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.
- (5) For any question concerning the technical content of the requirements in this AD, please contact:

AIRBUS – Airworthiness Office – 1IASA; E-mail: account.airworth-eas@airbus.com.

For full compliance please refer to:

<https://ad.easa.europa.eu/ad/2025-0068>

Appendix 1: Reference of the Affected Galleys

Note 2: Additional and removed series of affected galleys, compared to EASA AD 2024-0038 are highlighted in **bold**.

Series	P/N
601537	601537-XXXXXX
601539	601539-1, 601539-101, 601539-201 and 601539-301
601557	601557-1, 601557-301
601567	601567-1, 601567-20001, 601567-201, 601567-20101, 601567-20201, 601567-20701, 601567-21201, 601567-21301, 601567-21401, 601567-401
601853	601853-000301, 601853-000401
601854	601854-XXXXXX
601855	601855-000401, 601855-000601, 601855-000701, 601855-000801, 601855-000901, 601855-001001
601856	601856-000301, 601856-000501, 601856-000601, 601856-000701, 601856-000801, 601856-000901, 601856-001001, 601856-001101, 601856-001201, 601856-001501, 601856-001601, 601856-001701, 601856-001801, 601856-001901, 601856-002001, 601856-002101, 601856-002301, 601856-002401, 601856-002501, 601856-002601, 601856-002701, 601856-002801, 601856-002901, 601856-003001, 601856-003301, 601856-003401, 601856-003501, 601856-003801, 601856-003901, 601856-004001, 601856-004101, 601856-004301, 601856-004501, 601856-004601, 601856-004701, 601856-004801, 601856-004901, 601856-005001, 601856-005101, 601856-005201, 601856-005301, 601856-005401, 601856-005501, 601856-005701, 601856-005801, 601856-005901, 601856-006001, 601856-006101, 601856-006201, 601856-006301, 601856-006401, 601856-006501, 601856-007001, 601856-007101, 601856-007501, 601856-008001, 601856-011301, 601856-011601
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