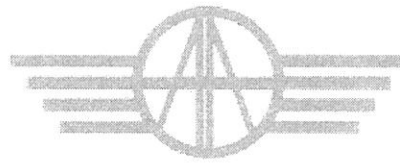




REPUBLIC OF ALBANIA



AUTORITETI I AVIACIONIT CIVIL

ALBANIAN CIVIL AVIATION AUTHORITY

AIRWORTHINESS DIRECTIVE

ACAA-DFS-AD-No.09

Issue: 01, Revision 00

Date: 12.07.2024

Approved by:

Maksim Et'hemaj

Executive Director of Albanian Civil Aviation Authority



0.1 Record of Amendments

The table below describes the dates and reason for the different amendments of the current procedure. A vertical black line on the left-hand side of the page identify the changes with the previous version.

Issue No.	Revision No.	Date	Amended by	Reason
01	00	12.07.2024	SAW	Initial Issue

0.2 Revision table

Page #.	Issue No.	Revision No.	Date	Edited by

1. Name of the AD:

EASA AD No.: 2024-0077, ATA 53 – Fuselage – Pressure Panel Longitudinal Beams – Inspection / Repair / Modification

2. Full Description of the AD:

This AD addresses cracks found around fasteners connecting the pressure panel with flexible brackets at fuselage frame (FR) 36 on Airbus A318, A319, A320, and A321 aeroplanes.

3. Issued and Effective Dates:

Issued: 19 March 2024

Effective: 02 April 2024

4. Full List of Aircraft Affected:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers (MSN), except those having Airbus modification (mod) 151574 embodied in production

5. Description of How It Can Be Resolved:

The AD mandates repetitive inspections and, if cracks are found, repair or modification of the pressure panel above longitudinal beams as per specified Airbus service bulletins (SB A320-53-1030, SB A320-53-1264, etc.).

For full compliance please refer to :

https://ad.easa.europa.eu/blob/EASA_AD_2024_0077_C1.pdf/AD_2024-0077_1