

REPUBLIC OF ALBANIA



ALBANIAN CIVIL AVIATION AUTHORITY

AIRWORTHINESS DIRECTIVE

ACAA-DFS-AD-No.006

Issue: 01, Revision 00

Date: 12.07.2024

Approved by:

Maksim #t'hemaj

Executive Director of Albanian Civil Aviation Authority



0.1 Record of Amendments

The table below describes the dates and reason for the different amendments of the current procedure. A vertical black line on the left-hand side of the page identify the changes with the previous version.

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0.2 Revision table

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1. Name of the AD:

EASA AD No.: 2024-0046, Time Limits / Maintenance Checks

Airworthiness Limitations Section Part 4 System Equipment Maintenance Requirements Amendment.

2. Full Description of the AD:

The System Equipment Maintenance Requirements (SEMR) for the Airbus A320 family aeroplanes, which are approved by EASA, are currently defined and published in the Airbus A318/A319/A320/A321 ALS Part 4 document. These instructions have been identified as mandatory for continued airworthiness. Failure to accomplish these instructions could result in an unsafe condition.

EASA previously issued AD 2022-0102 to require accomplishment of all maintenance tasks as described in ALS Part 4 at Revision 08. Since that AD was issued, Airbus published the ALS, as defined in this AD, to introduce new and/or more restrictive tasks. For the reason described above, this AD supersedes EASA AD 2022-0102 and requires accomplishment of the actions specified in the ALS...

3. Issued and Effective Dates:

Issued: 19 February 2024 Effective: 04 March 2024

4. Full List of Aircraft Affected:

Airbus A318-111, A318-112, A318-121, A318-122

Airbus A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A319-151N, A319-153N, A319-171N

Airbus A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A320-251N, A320-252N, A320-253N, A320-271N, A320-272N, A320-273N

Airbus A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231, A321-232, A321-251N, A321-251NX, A321-252NX, A321-252NX, A321-253NX, A321-271NX, A321-272NX

5. Description of How It Can Be Resolved:

Maintenance Tasks and Replacement of Life-Limited Parts:

Replace each component before exceeding the applicable life limit. Within the thresholds and intervals, accomplish all applicable maintenance tasks. Corrective Actions:

If discrepancies are found during the accomplishment of any task, before the next flight, accomplish the applicable corrective action(s) in accordance with the applicable Airbus maintenance documentation. If a detected discrepancy cannot be corrected using existing Airbus instructions, before the next flight, contact Airbus for approved instructions and accomplish those instructions accordingly. AMP Revision:

Within 12 months after the effective date of this AD, revise the Aircraft Maintenance Programme (AMP) by incorporating the limitations, tasks, and associated thresholds and intervals described in the ALS, as applicable to the aeroplane model and depending on aeroplane configuration.

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Credit:

If, before the effective date of this AD, the AMP has been revised to incorporate the maintenance tasks and life limitations as specified in a previous revision of ALS Part 4, that action ensures the continued accomplishment of those tasks and limitations.

For an aeroplane to which that AMP applies, it is acceptable to accomplish the new and/or more restrictive tasks and limitations as specified in the ALS, as applicable to aeroplane model and depending on aeroplane configuration, within the compliance times as specified in the ALS to comply with paragraph (1) of this AD.

For that AMP, it is acceptable to incorporate the new and/or more restrictive tasks and limitations as specified in the ALS, as applicable to aeroplane model and depending on aeroplane configuration, into the AMP to comply with paragraph (3) of this AD.

Recording AD Compliance:

When the AMP of an aeroplane has been revised as required by paragraph (3) or (4) of this AD, that action ensures continued accomplishment of the tasks as required by paragraphs (1) and (2) of this AD for that aeroplane. Consequently, after revising the AMP, as required by paragraph (3) or (4) of this AD, it is not necessary that the accomplishment of individual actions is recorded for demonstration of AD compliance on a continued basis.

Ref. Publications:

Airbus A318/A319/A320/A321 ALS Part 4 Revision 09 dated 06 November 2023. The use of later approved variations or revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.

This AD was posted on 10 January 2024 as PAD 24-001 for consultation until 07 February 2024. No comments were received during the consultation period.

Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.

Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part, or appliance not affected by this AD, can be reported to the EU aviation safety reporting system. This may include reporting on the same or similar components, other than those covered by the design to which this AD applies if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC), or other modification.

For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – 1IASA; E-mail: account.airworth-eas@airbus.com.

For full compliance please refer to:

https://ad.easa.europa.eu/blob/EASA_AD_2024_0046.pdf/AD_2024-0046_1

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